

Subsystem Requirements

Introduction

Subsystems

- Actuation
- Formation Control
 - Control
 - Metrology
 - Requirements
 - Trades
 - Design
 - Issues
 - Budget
- Estimates
- Electronics
- Structure/Power

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- Determine separation distance
 - Goal: sense to one-tenth the control tolerance
- Determine relative attitude (direction) of vehicle to an angular tolerance dependent on angular controllability
 - Goal: sense to one-tenth the control tolerance
- Full field of view : 360 degrees in two dimensions
- Sensing presence of other vehicles to a distance compatible with test facilities

Trades

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Sonic and IR based system

- Similar to SPHERES metrology
- Expected improvement in performance due to 2D operation
- New technology should eliminate some accuracy of the errors encountered by SPHERES
- Low refresh rate will require the use of gyros and accelerometers

Design

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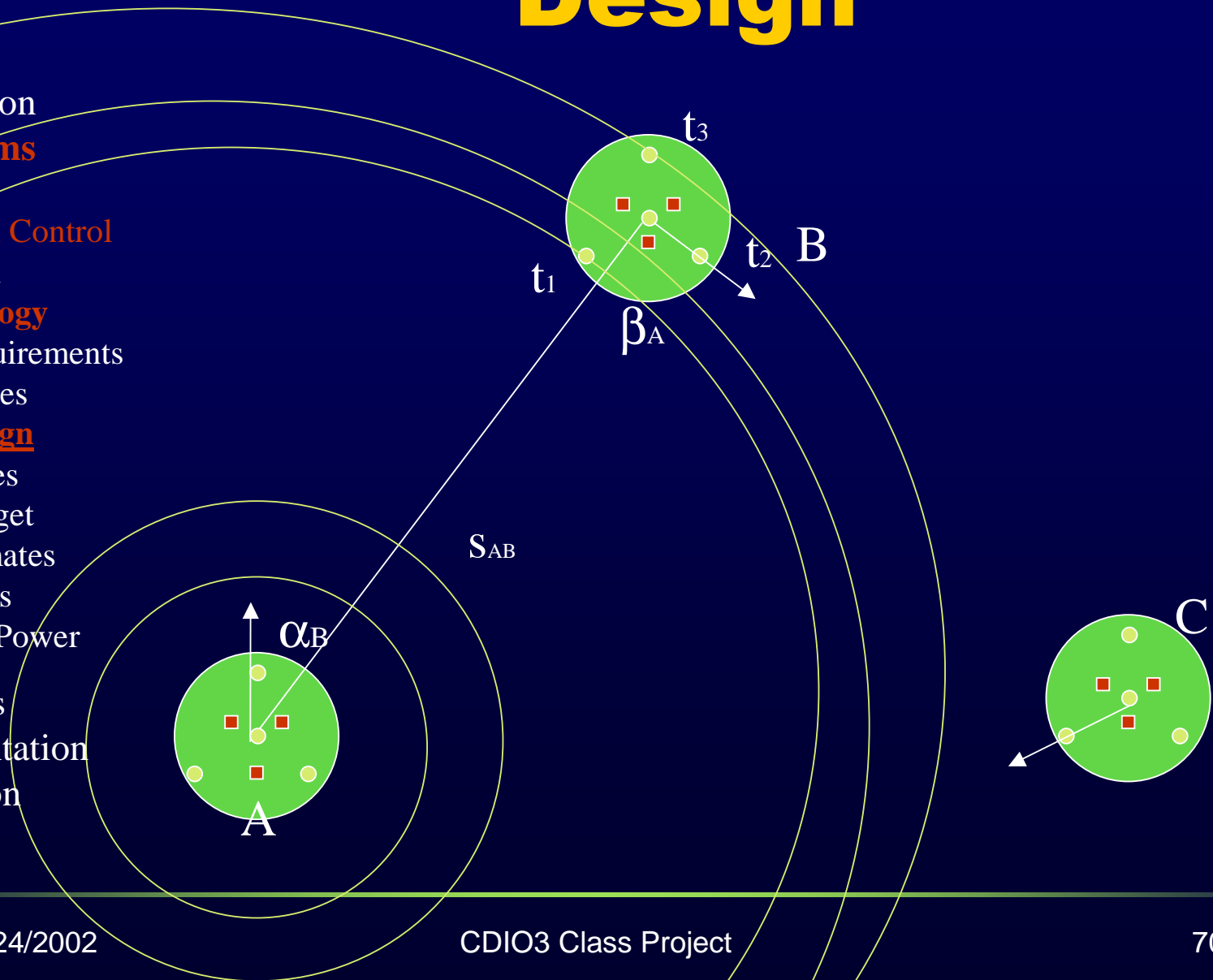
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- 1 ultrasonic omni-directional transmitter (Mimio)
- 3 ultrasonic omni-directional receivers (cones)
- 3 IR receivers & 2 transmitters
- 1 rate gyro
- 1 2-axis accelerometer

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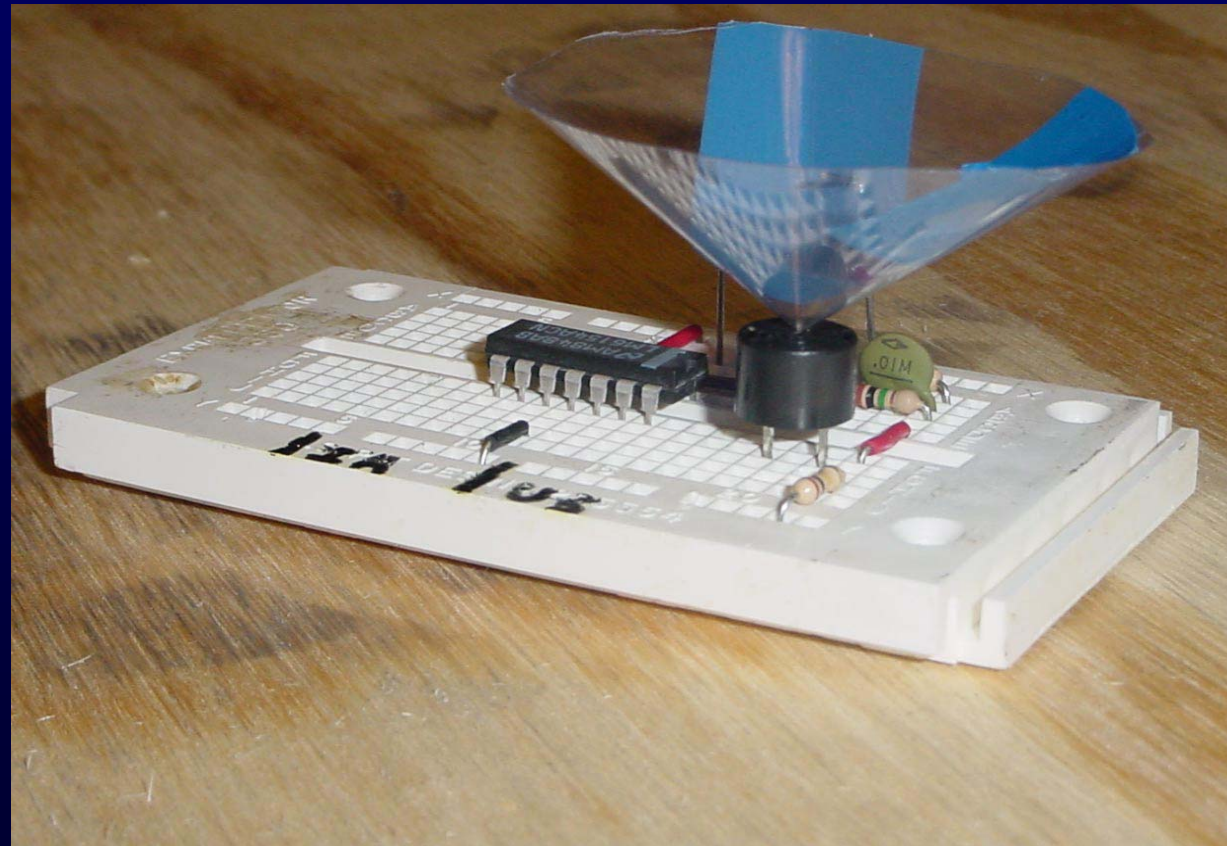
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- System relies only on distance readings
- Uses data from all three sensors
- Calculates relative attitude and distance directly to center of vehicle

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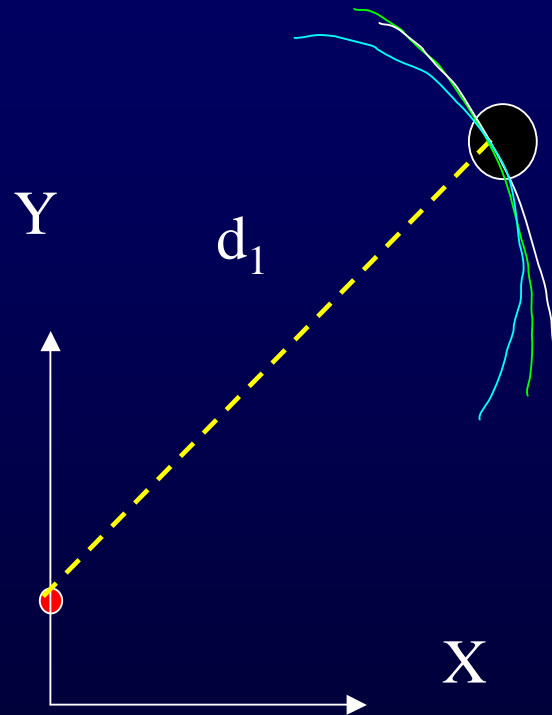
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- Origin at center of vehicle
- x_i and y_i are position of the sensors

$$(x + x_1)^2 + (y + y_1)^2 = d_1^2$$

$$(x + x_2)^2 + (y + y_2)^2 = d_2^2$$

$$(x + x_3)^2 + (y + y_3)^2 = d_3^2$$

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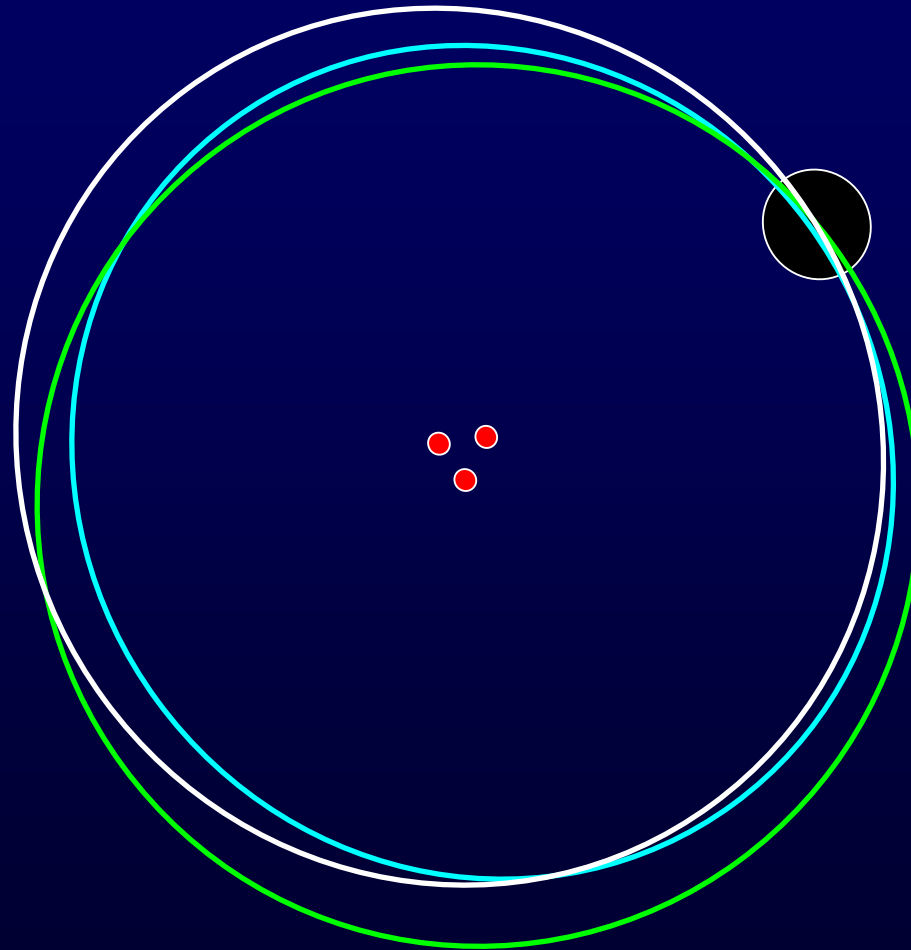
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- Omni-directional sonic sensors
 - Hand made cones added to current sensors
- Effect of magnetic forces
- Range and accuracy
- Refresh rate
 - Sound signal leaving the testing area
 - Rate gyros and accelerometers

Budget Estimates

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Part	Cost (\$)	Mass (kg)	Power (W)
Sonic (1+3)	70	0.05	0.3
IR (2+3)	30	0.04	0.25
Gyros	1200	0.06	0.36
Accelerometers	1200	0.05	0.18
Total (per vehicle)	2500	0.20	1.09
Total (system)	7500	0.60	3.27

Communications

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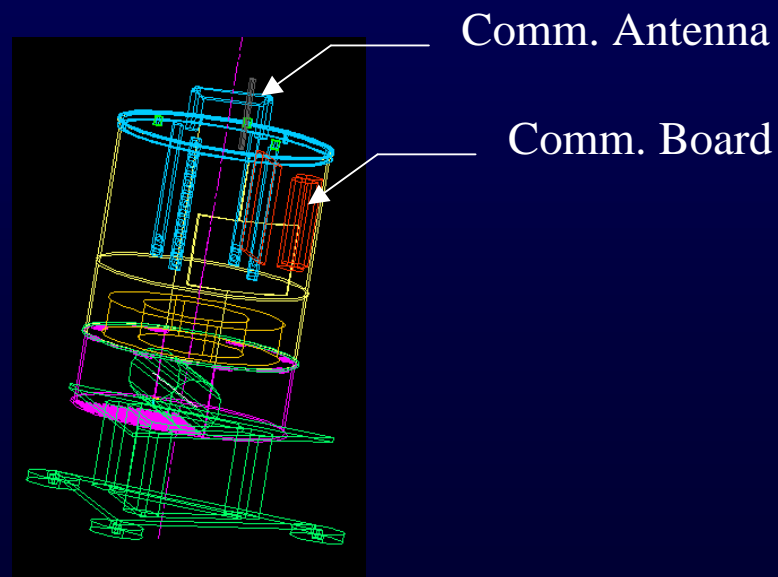
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Jennifer Underwood



Communications

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● “The technology employed in transmitting messages”

- Architecture
- Hardware
- Software
- Interfacing

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- Send information and instructions automatically from vehicle to vehicle
 - Control and metrology purposes
- Send information and instructions on command from ground to vehicle
 - Begin preprogrammed tests
 - Emergency intervention procedures
- Send flight health data to “ground” operator
- Have no protruding antennae that might interfere with dynamics

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Trades

● Hardware

- Processor board (chosen by avionics → TT8)
- Transceiver

● Architecture

Transceiver Metrics

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- EM rejection (frequency)
- Bandwidth/data rate
- Weight
- Ease of interface
- Size
- Cost
- Power consumption
- Range

Transceiver Trades

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- Radio Frequency (RF) vs.. Wireless LAN
 - Avg cost of LAN > avg cost of RF
 - LAN requires a base station (\$\$)
 - Size and weight of LAN > RF
 - LAN bandwidth, range > RF
 - Power drain of LAN < RF
 - Both have capacity to reject EM (high frequencies) and are easily interfaced
- Choice: RF

RF Trades

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	AC5124C-10	RF Monolithics (DR3000-1)
Size	2.65"x1.65"x0.20"	1"x1.5"
Power	0.35 W	0.04 W
Frequency	2.402 –2.478GHz	916.5 MHz
Weight	0.02 kg	Hardly any, < AC5124C-10
Data rate	115.2-882 kbps	115.2 kbps
Range	91m indoors	Short-range wireless
Ease of Interface	Relatively Easy	Relatively Easy
Cost	~\$245	\$35
Availability	Company in Europe	DR1012 avail from SPHERES for prototyping
Complexity	OEM kit, not familiar	Development kit ready, familiar to staff, students

Transceiver Selection

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● Preliminary final product:

- DR 3000-1
 - Sufficient EM rejection
 - Familiarity
 - Power drain

● Prototyping product:

- DR1012
 - Availability
 - Familiarity

Architecture Trades

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● Sequential

- Pass token to determine who talks and to whom
- Hub makes calculation

● Simultaneous

- Pass token to determine who talks to everyone
- All vehicles make calculations

● Hybrid

- Combination of Sequential and Simultaneous

● Reliability deciding factor

Architecture Trades

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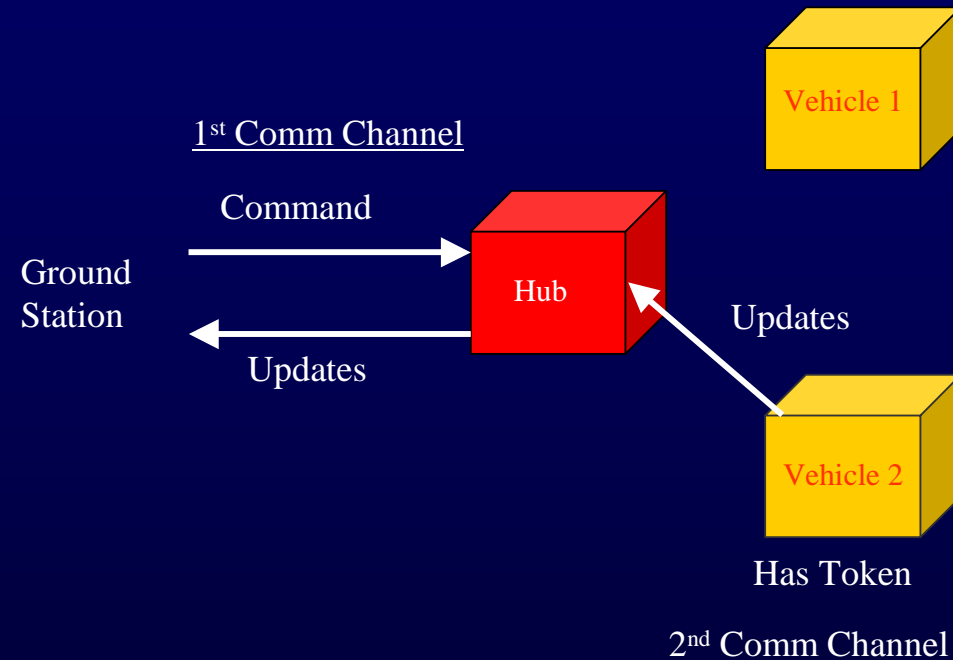
Operations

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Sequential

- Vehicle 2 talks to Hub
- Token passed to Vehicle 1
- Vehicle 1 talks to Hub
- Hub makes control calculations and sends commands to Vehicle 1 and Vehicle 2



Architecture Trades

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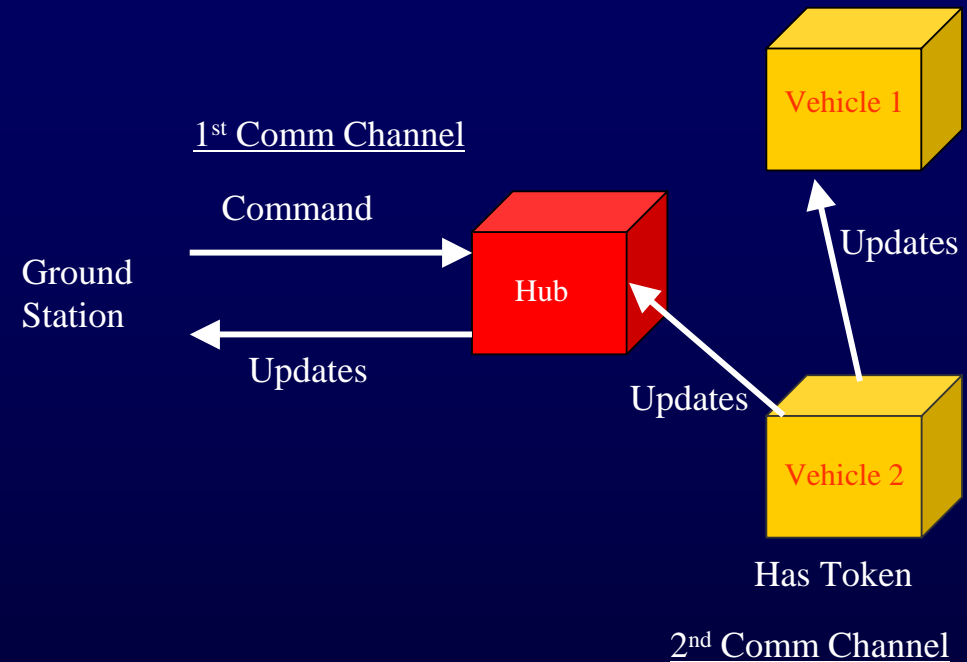
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● Simultaneous

- Vehicle 2 talks to Hub and Vehicle 1
- Token passed to Vehicle 1
- Vehicle 1 talks to Hub and Vehicle 2
- Token passed to Hub
- Hub talks to Vehicle 1 and Vehicle 2
- Each vehicle makes control calculations



Architecture Trades

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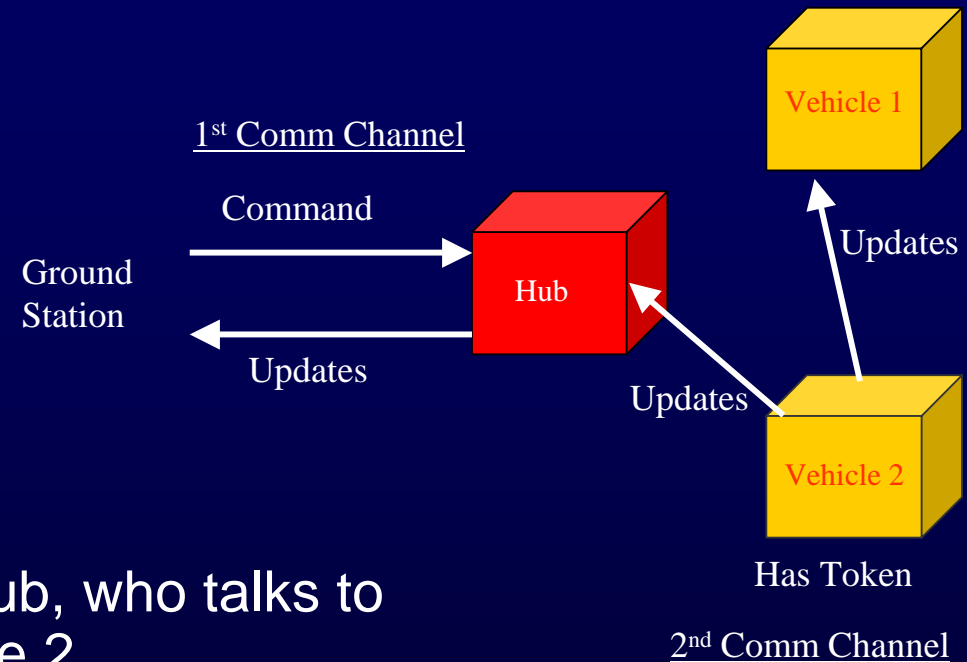
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● Hybrid

- Vehicle 2 talks to Hub and Vehicle 1
- Token passed to Vehicle 1
- Vehicle 1 talks to Hub and Vehicle 2
- Token passed to Hub, who talks to Vehicle 1 and Vehicle 2
- All vehicles make control calculations but only Hub determines commanded control vector → sends commands to Vehicle 1 and Vehicle 2, ground listens in



Architecture Trades

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● Sequential

- Requires excess code, bandwidth (BW)
- Reliable from control point of view

● Simultaneous

- Easy to implement
- Not reliable from control point of view

● Hybrid

- Reliable and versatile from control and comm point of view
- Increased bit rate required, excess code, BW

● Design Choice: Hybrid

Design

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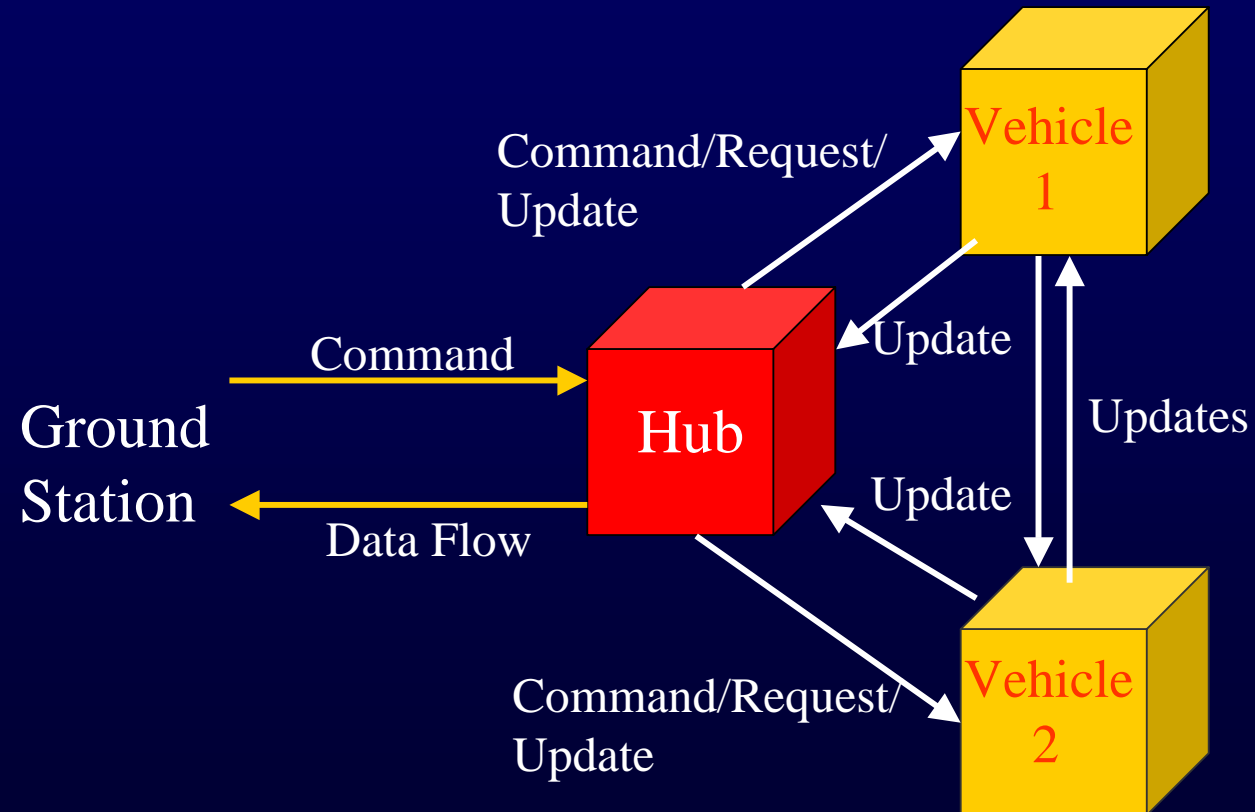
• Avionics

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● Design Considerations

- Comm channel usage
- Transmission rates
- Data framing
- Error detection/correction
- Channel coding
- Automatic Repeat Request (ARQ) protocols

Data Framing

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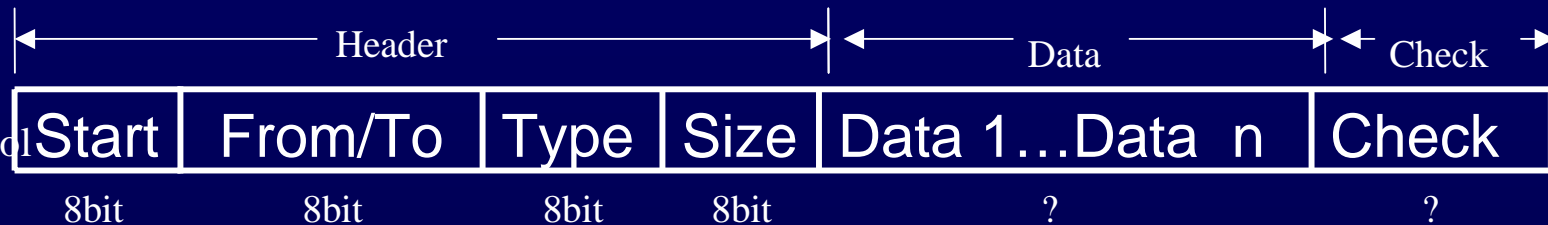
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● Data framing is essential

- Who to send info to
- Who sent info (Hub, Vehicle 1, Vehicle 2)
- Type of data
- Size of data packet
- Error checking

● Ease of transmitting chunks (1 byte)

Channel Usage

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● Cluster Comm Channel

- 6 variables in state vector (Control)
- 4 variables to actuators (EM, RWA)
- About 800 bits/cycle for control
- About 500 bits/cycle for health updates

● Ground Link Comm Channel

- Undefined requirements
- On the order of 400 bits per complete cycle

Transmission Rates

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- How frequently to measure systems/states?
 - Control runs at 50 Hz
 - Health max set at 10 Hz
- Therefore, we can estimate the transmission rate required:
 - $800 \text{ bits/cycle} * 50 \text{ cycles/sec} + 500 \text{ bits/cycle} * 10 \text{ cycles/sec} = 45\text{kbps}$

Current Capabilities

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- TT8 to TT8 communication
 - Currently connected through UART channel
- Two byte transmission
 - Send and receive between two TT8's, one direction only
 - High-Byte, Low-Byte
 - [1 1 1 1 1 1 1 1] [1 1 1 1 1 1 1 1] = 1111111111111111
High Byte | Low Byte

Issues

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- Processor/transceiver shielding
- Reliability
 - EM resistance
 - Error probability
- Communication channel load

Budgets Estimates

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Part	Cost (\$)	Mass (kg)	Power (W)
Transceiver	\$35 each	0.02	0.04
Miscellaneous parts	\$100	0.1	-
Replacements/repairs	\$70	-	-
Total (per vehicle)	\$275	0.24	0.08
Total (ground station)	\$275	0.24	0.08
Total (system)	\$1100	0.96	0.32

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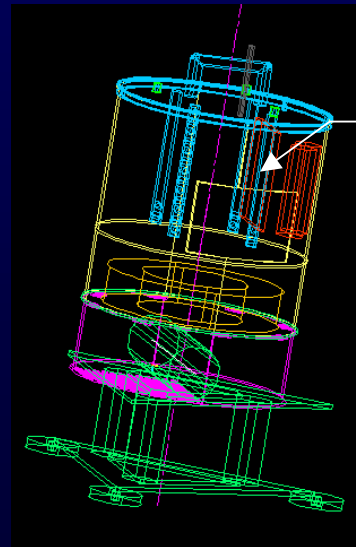
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Avionics



Avionics Board

Subsystem Requirements

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- Manage timing and resources for all subsystems
- Run control loop in real-time
 - Inputs, calculations, outputs
- Administer preprogrammed tests
- Be easily programmable
- Stay within system budgets

Trades - Metrics

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- Processing speed
- Interfaces (I/O)
- Data storage (RAM/ROM)
- Constraint considerations
 - Cost
 - Size and mass
 - Ease of use

Trades - Metrics Details

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● Processor speed

● I/O capabilities

● Digital vs.. analog

● Subsystem input and output needs not necessarily the same

● RAM and ROM capacity

● Flash memory

● Power consumption

● Cost

Trades - Interfaces

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The avionics team must interface with all subsystems.

INPUTS

● Metrology:

- 3 Ultrasonic sensors (digital)
- 1 IR timing (digital)
- 1 rate-gyro (analog)
- Possibly 2 accelerometers (analog)

● Communication

- Inter-vehicle data and instructions (digital)

● Health Indicators (digital)

Trades - Interfaces

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OUTPUTS

● Communication

- Requests and commands (digital)

● Actuators

- 3 for Y-pole magnet (analog)
- 1 for RW (analog)

● Metrology

- 1 for ultrasonic transmitter (digital)
- 1 for IR timing transmitter (digital, split)

Trades: Computer Comparison

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<u>Feature</u>	<u>Needs</u>	<u>TT8</u>	<u>C40</u>	<u>6701</u>
Speed	50 Hz	16 MHz 4 MIPS	50 MFLOPS	167 MHz 1 GFLOPS
I/O	4D, 1A in 1D, 4A out	25D I/O 8A, 14 time	D parallel 64 I/O	N/A
RAM	~16 kB	256kB	16 MB	16 MB
ROM	~8kB	256kB	640 kB	512 kB
Power	Low	1.8 Watts	N/A	N/A
Cost	Low	(~\$500)	Custom	High

Trades: other considerations

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- Other Computer Considerations
- Size/weight satisfactory for structure?
- Available/Replaceable?
- Easy to use?
- Expandable?

Design: Hardware

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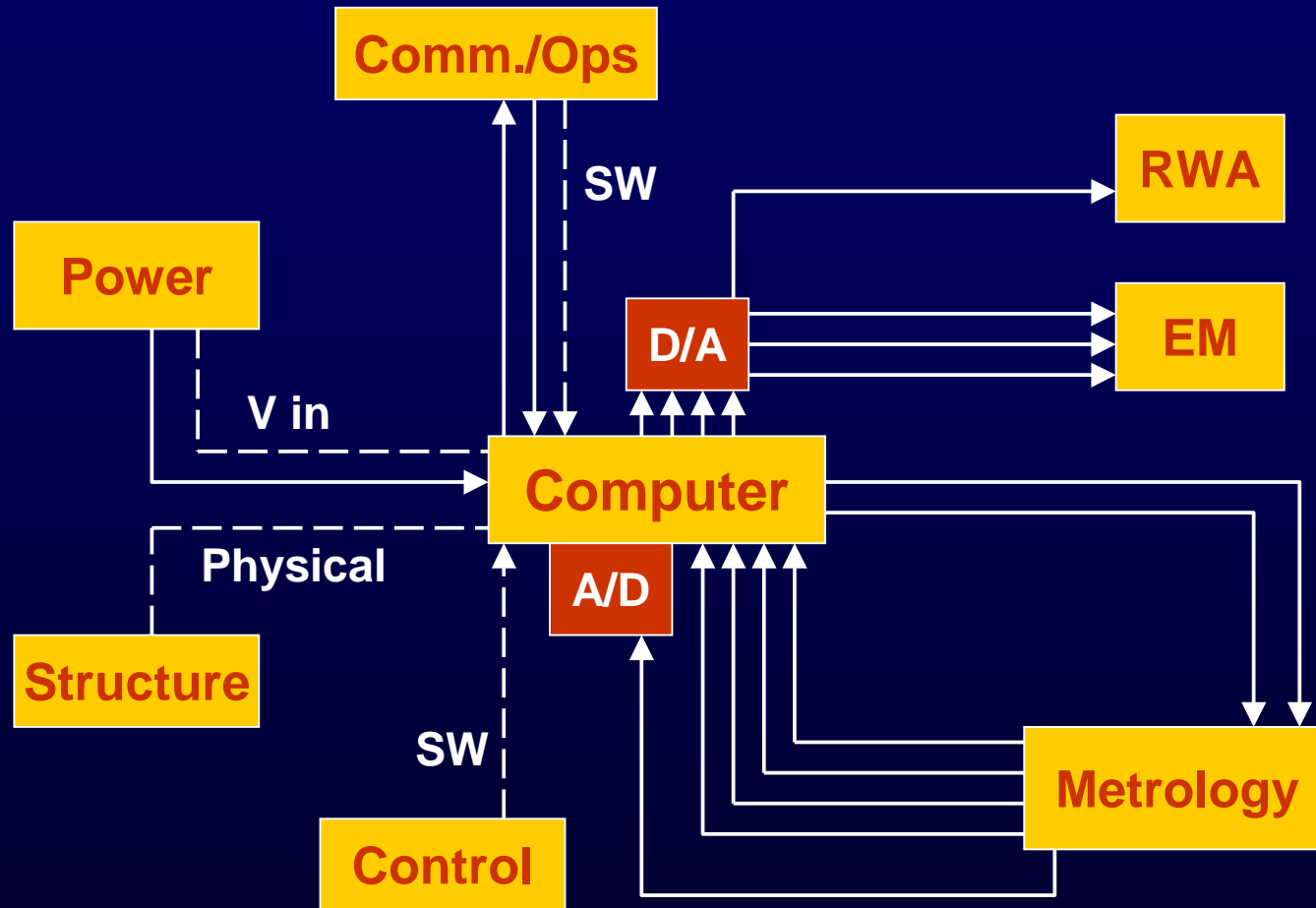
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Design: Software

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- Language: C
- Coding environment
 - Creating: Metroworks CodeWarrior
 - Loading to vehicle: Motocross
- Procedures
 - Control loop
 - Metrology updates
 - Matrix calculations
 - Actuation commands
 - Test programs
 - Health, test data reports

Issues

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- Software is the high-risk item for the avionics subsystem.
- Preliminary Test Prototyping Design to find complications early.
 - Develop a clock-interrupt
 - Blink an LED
 - Signal Reproduction via PWM
- Preliminary Prototype to be completed by end of semester.

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● Mass

- 0.028 kg per tattletale
 - May need two
- 0.028 per subsystem circuit board

● Power

- Highest power draw: two main computers
- Power required: 3.6 Watts

● Cost

- Per-vehicle needs, system-wide extras
- \$500 total allocated for TT8 repairs
- \$6500 for circuit boards (power, comm., controls, metrology)

Budget Estimates (cont.)

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Part	Cost (\$US)	Mass (kg)	Power (W)
TT8 (x2)	500	0.057	3.6
Boards	6500	0.113	?
Total (vehicle)	1060	0.113	?
Total (system)	7000	0.339	?

Power

Amy Schonscheck

Introduction

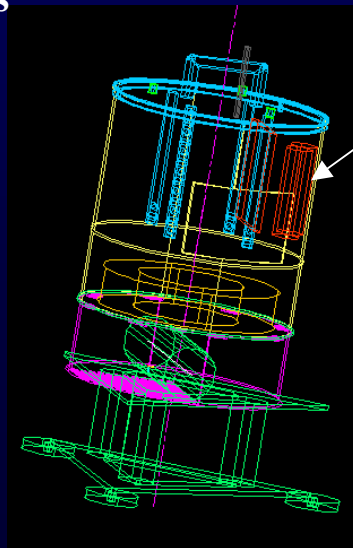
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- No external umbilicals
 - On-board power supply
- Provide sustainable power for 30 minutes
- Use a renewable or rechargeable energy source

Power Trades

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- Solar power vs.. batteries
 - Power requirements too demanding for solar power
 - Rechargeable batteries the best option
- Choice of battery chemistry

Battery Selection

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- For expected voltage and current requirements, viable options include:
 - Lithium Ion (Li-ion)
 - Nickel Cadmium (NiCd)
 - Nickel Metal Hydride (NiMH)

Battery Selection

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- Li-ion:
 - High energy density
 - Low discharge rate (2 Amps max.)
- NiCd:
 - Adequate discharge rate
 - Low efficiency at high current draws
- NiMH: → Final choice
 - High energy density
 - Fast discharge rate
 - Efficient even at high current draw

Power Design

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Subsystem power estimates:

Subsystem	Voltage (V)	Power (W)
Avionics: Tattletale (x 2?)	9-12 (each)	2 (each)
Metrology: Accelerometers	8-30	0.18
Metrology: Gyros	12-18	0.36
Metrology: Transmitters/Receivers	3	0.066
Comm/Ops: (shares Tattletale)	9-12	0.1
RWA: Reaction Wheel	28?	13
EM: Electromagnet	12 (min.)	60-70
	TOTAL:	~ 90 W (max)

Power Design

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- Total current draw: 7 Amps (current design)
- Operation time: 30 minutes (min.)
- Total energy: 3.5 Ah
- Candidate battery: Panasonic HHR200SCP
 - Voltage - 1.2 V
 - Capacity - 1.9 Ah
 - Weight – 42 g
 - Dimensions: 23 mm diameter, 34 mm height

Power Design

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- Candidate battery architecture:
 - 1 “Pack” = 15 batteries x 1.2 V (wired in series)
 - 18 V
 - 1.9 Ah
 - 2 Packs wired in parallel
 - 18 V
 - 3.8 Ah
 - Sufficient voltage, current for the system

Power Design

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- Voltage Regulators used to step down voltages
 - 9-12 V : Tattletale processors
 - 5 V : Transmitters/Receivers
 - Gyros, accelerometers may have built-in voltage regulators
- Switchmode amplifier used to control current through electromagnet
 - Commercially available

Power Flowchart

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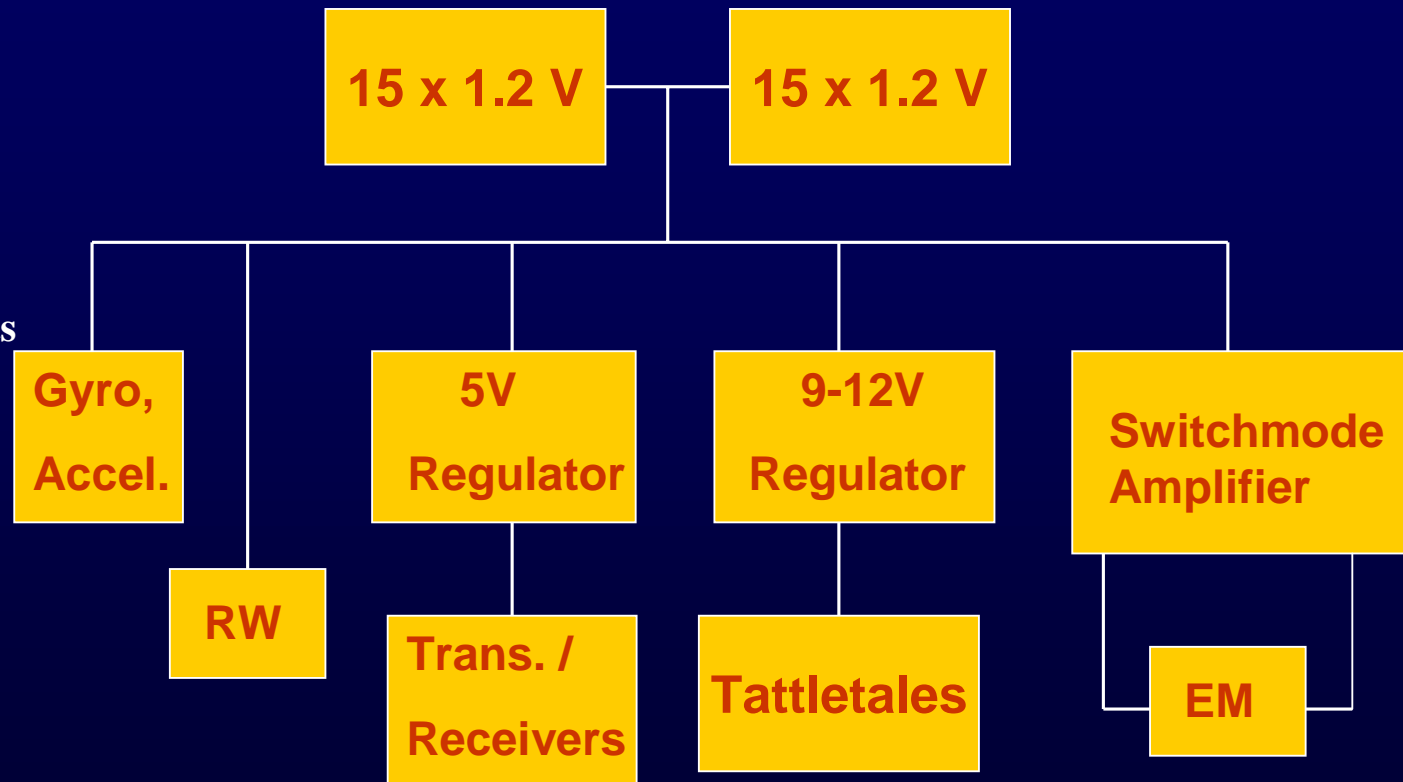
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Issues

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- NiMH batteries have strict charge/discharge limits
 - Overcharging decreases performance
 - Rapid discharge → produces heat
- Safety precautions
 - Avoid excessive heat
 - Avoid contact with water

Issues

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- Possible greater power requirements
 - Current design allows 7 amps
 - EM may require up to 10 amps
 - May require more batteries
 - Possibly switch to next higher battery model (much higher mass)
- Charging takes ~ 1.2 hours
 - Two or three complete battery sets needed (per vehicle) → higher cost

Budget Estimates

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● Mass:

- 30 batteries
- 42 g each
- Total battery weight: 1.26 kg
- Additional components: ~ 50 g (regulators, amplifier)

Budget Estimates

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● Cost:

- Switchmode amp: ~\$65
- Voltage regulators: ~ \$60
- Batteries: \$400 - \$500

● Power:

- Supply 100% of EMFFORCE power needs

Structure

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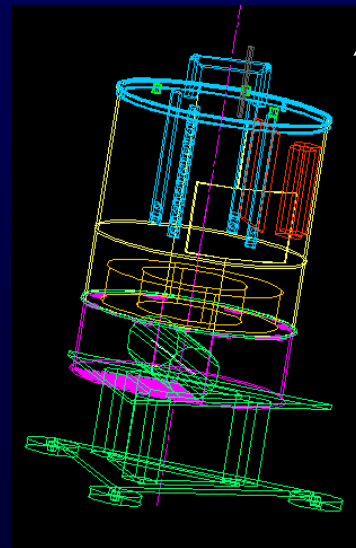
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Structure Requirements

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- Vehicle Casing:
 - Provide physical interfacing capability
 - Prevent damage in case of collision
 - Thermal considerations due to magnet heating
- Magnetic Shielding:
 - Protect electronics hardware from magnetic interference
- Air Carriage:
 - Provide adequate cushion height for vehicle mass

Geometric Overview

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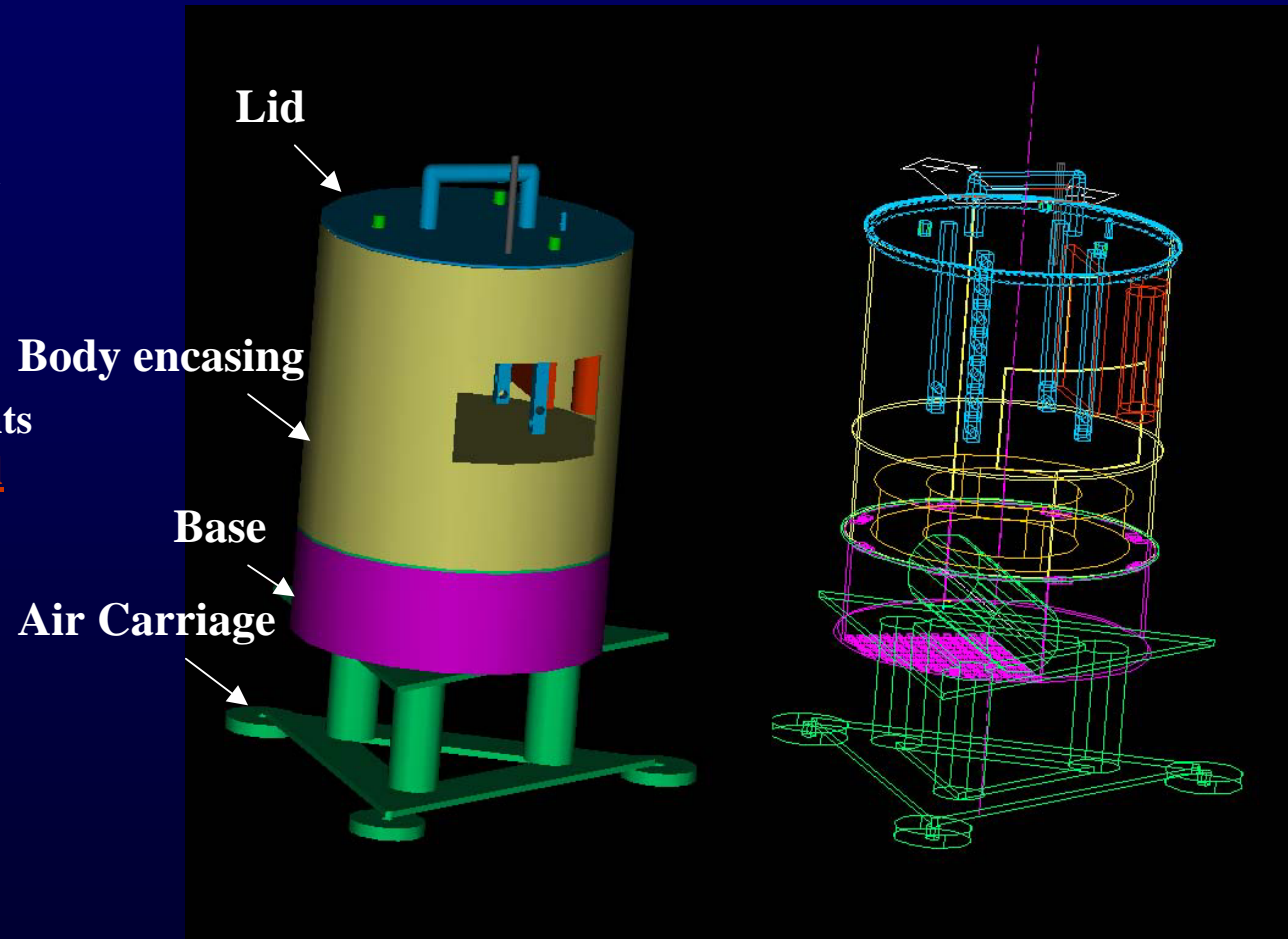
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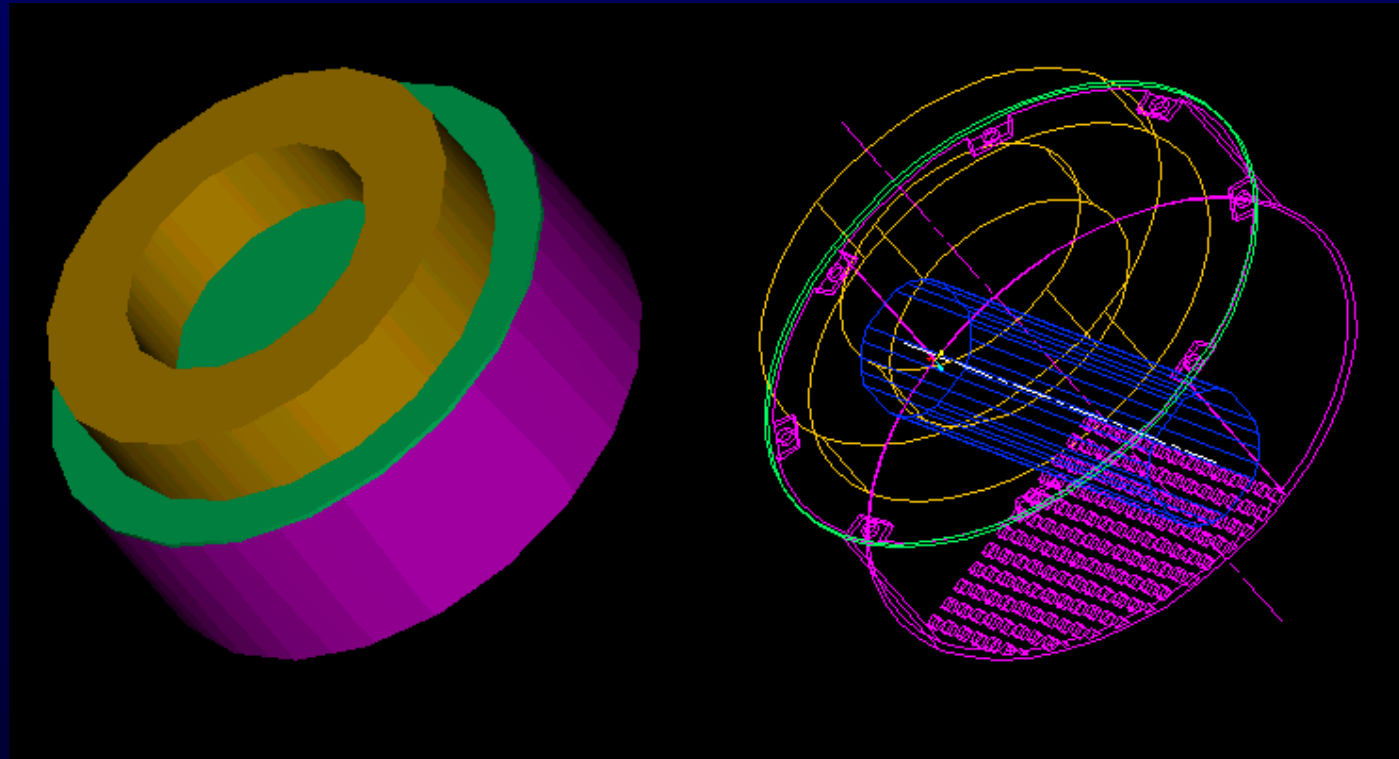
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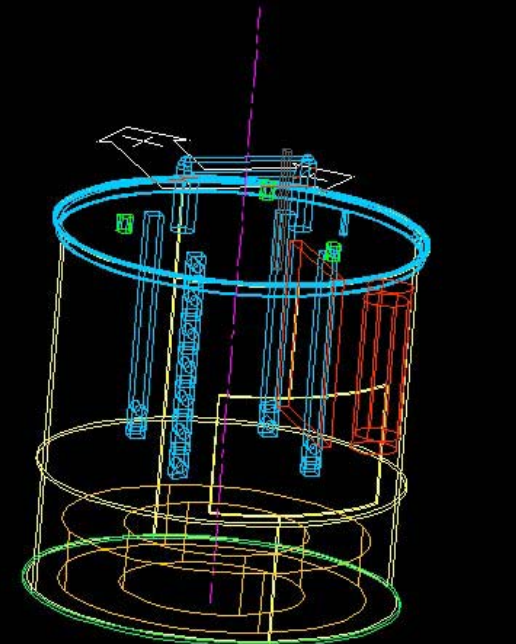
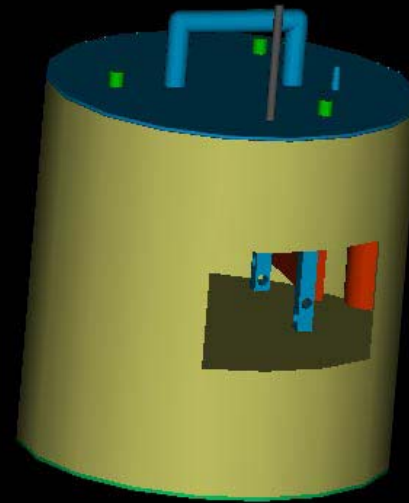
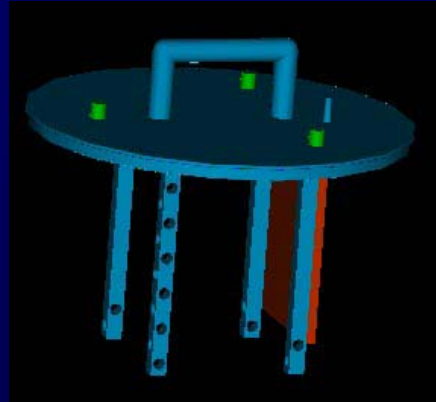


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Shielding

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- Sample kit includes various materials of different properties.
- Determine functional material at least mass.
- Conduct tests using electronics/ electromagnets, and test shielding material.

Air Carriage: Trades

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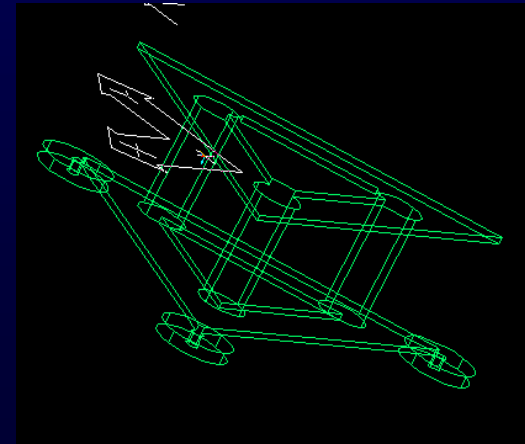
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- Fabricate vs.. Off-the-Shelf
 - Cost, Efficiency
- Tanks vs.. Compressors
 - Cost, Power, Mass
 - Infinite(?) air supply



Air Carriage: Design

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- Objective: For a given weight accurately predict and obtain a maximum air cushion thickness
- Design variables
 - Supply pressure
 - Puck radius
 - Supply orifice

Air Carriage: Model Assumptions

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- Linear radial pressure distribution for single, central orifice

$$P(r_p) = P_s - (P_s - P_a) \frac{r_p}{R_p}$$

- Possible compressible effects near aperture

Air Carriage: Lubrication Theory

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● Thin film:

- $h \ll R_p$
- very low Reynolds Number

$$\text{Re} = \frac{\rho_a U c}{\mu_a} \propto \frac{\rho_a R_p U_p}{\mu_a} \left(\frac{h^2}{R_p^2} \right)$$

● Similarities to Couette and Poiseuille flows: parabolic velocity distribution

Air Carriage: Next Steps...

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- Lubrication flow solver
 - Pressure gradient
 - Flow Velocity
 - Load
 - Cushion thickness
- Assess compressible behavior
- Assess puck designs